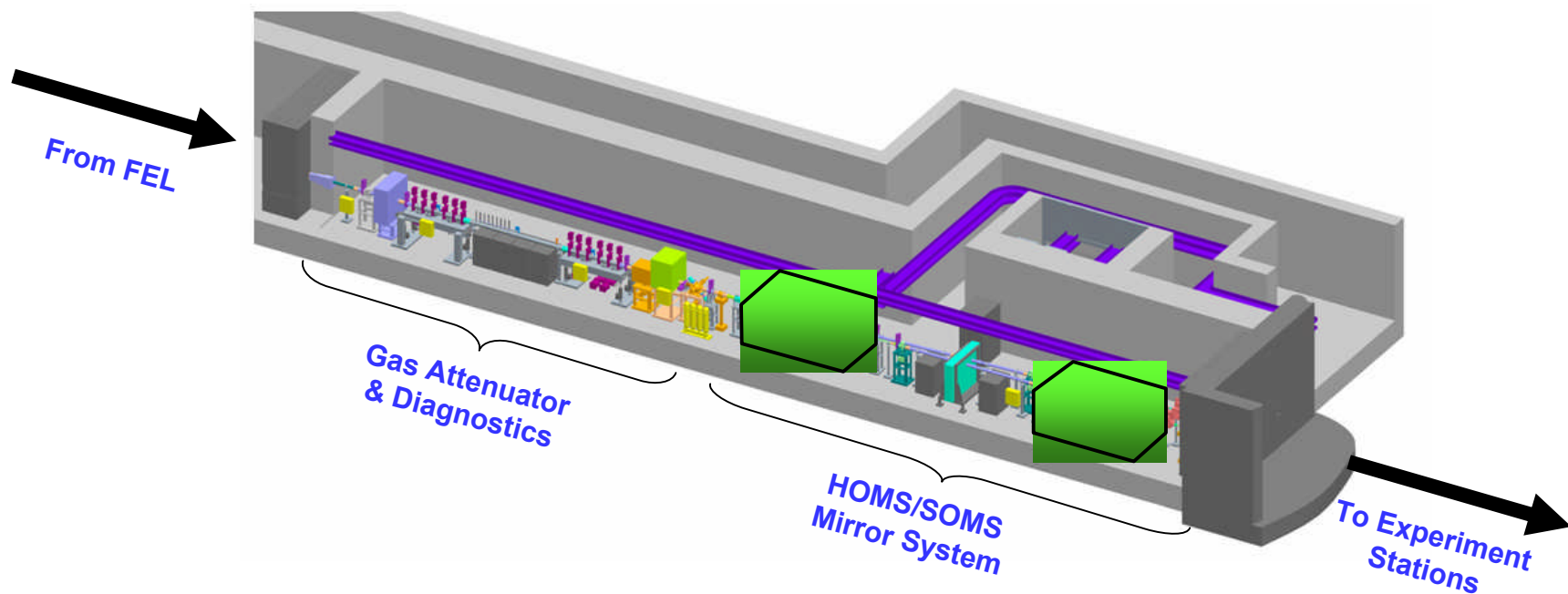


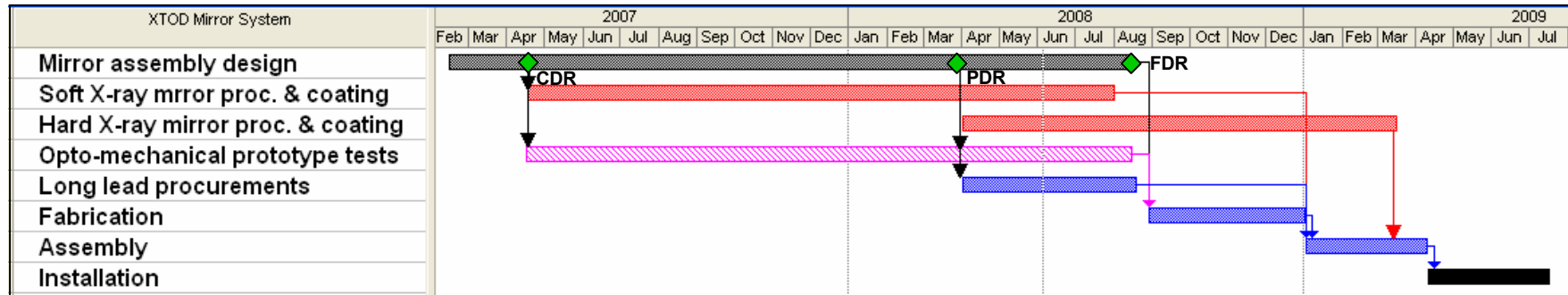
Hard and Soft X-ray Mirror System Design Status



This work performed under the auspices of the U.S. Department of Energy by Lawrence Livermore National Laboratory under Contract DE-AC52-07NA27344.



Mirror system testing & final design drawings are nearly complete



- **Prototype tests focused on two high technical risk areas:**
 - Measuring & maintaining HOMS tangential figure
 - Measuring & maintaining HOMS pointing stability
- **Requirements have been demonstrated for both these areas**



HOMS and SOMS share a common design approach: HOMS requires a few additional features

Chamber & mirror

Pointing & centering

Installation alignment plates

Support Pedestal

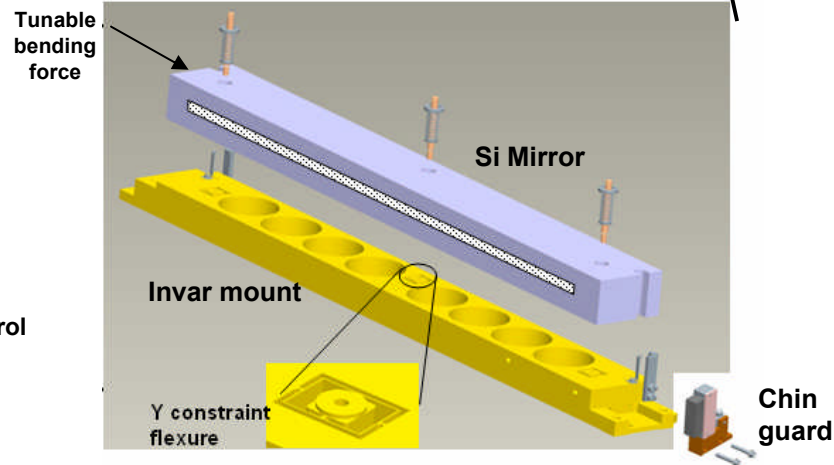
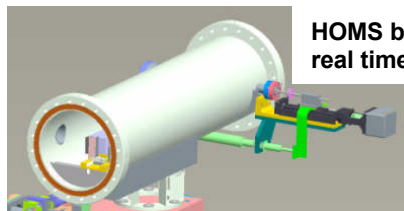
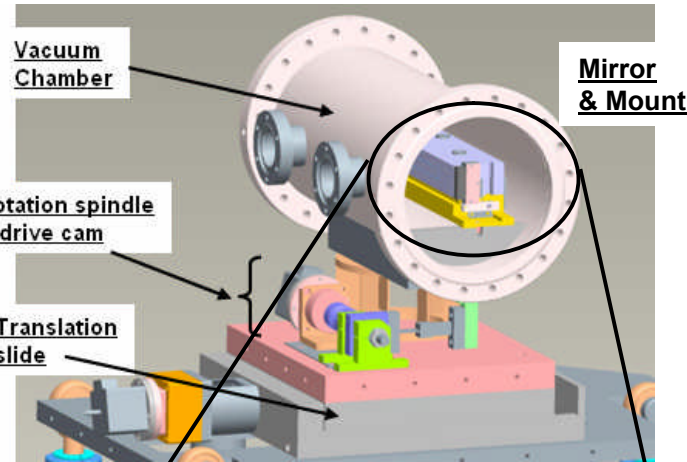
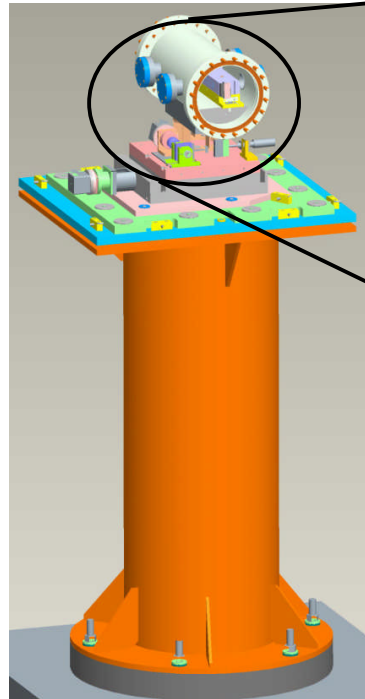
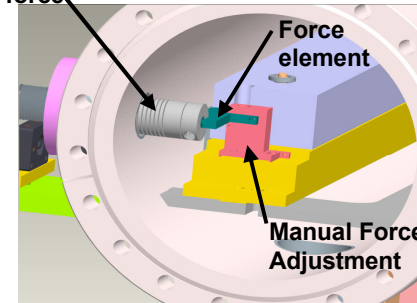


Figure errors predicted by analysis have been verified in prototype tests

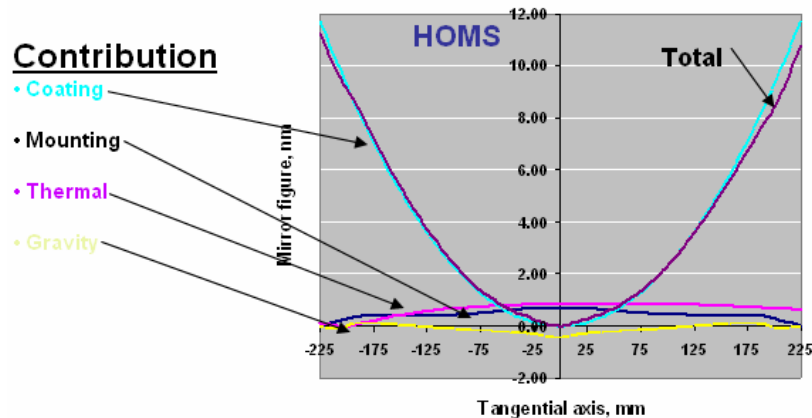
- Figure budget limits “power” imparted upon reflection

Mirror Flatness Requirement			Figure error budget, nm				
Axis	"Best Fit" Radius, km	Peak to Valley, nm	Mounting	Thermal	Coating	Gravity	Fabrication
HOMS/SOMS Sagittal	>~ 3	< 5	<<1	<<1	<<1	<<1	<1
SOMS Tangential	>188	<20	<1	<1.5	<4	<-1	<10
HOMS Tangential	>2600	<9	<1.5	<1.5	<12	<-1	<20

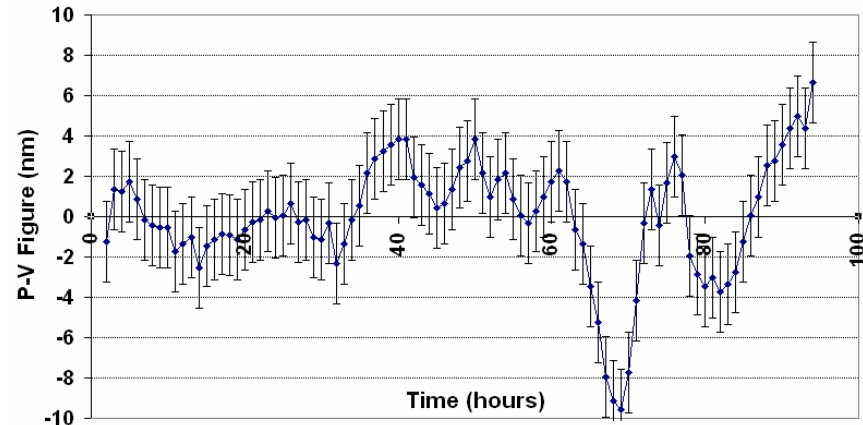
Remotely controlled spring force



- Finite element analysis guided design optimization

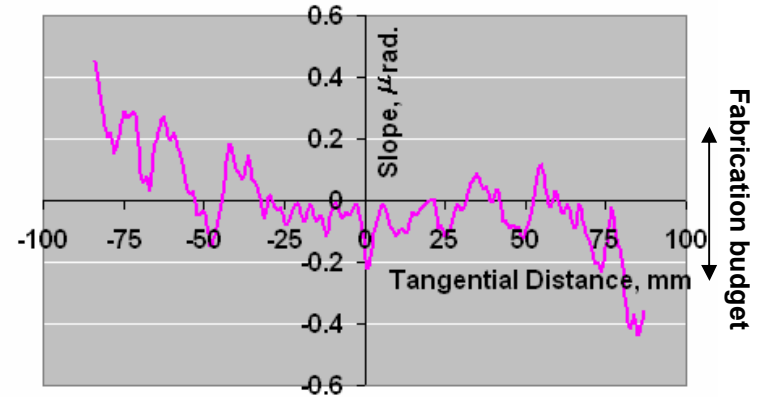
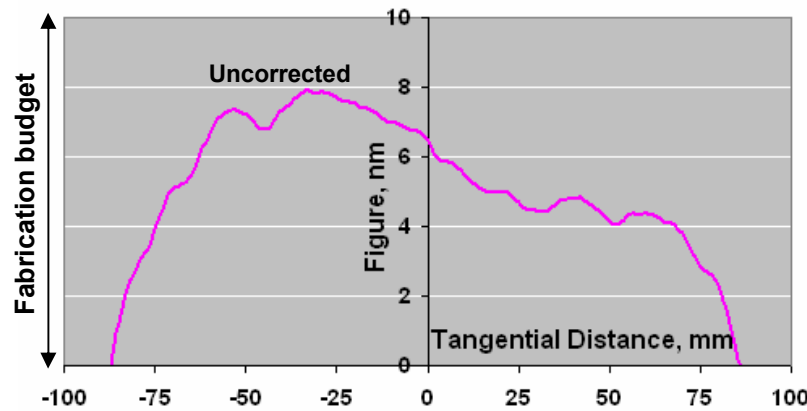
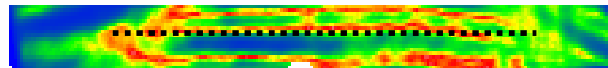


- Mirror figure stability after 150 nm bend: +/- 1 C° room air



SOMS mirror #1 tangential (horizontal) figure has been measured (transmission flat correction not yet applied)

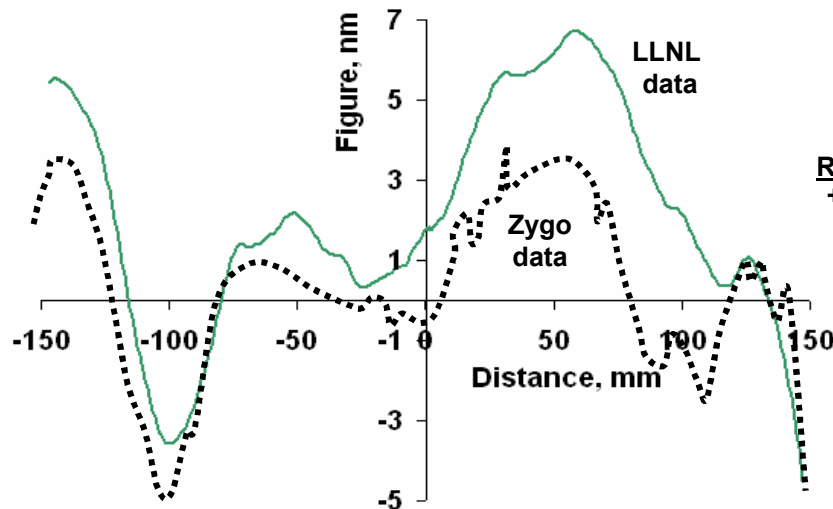
SOMS #1 mirror figure and slope



We recently calibrated our interferometer transmission flat

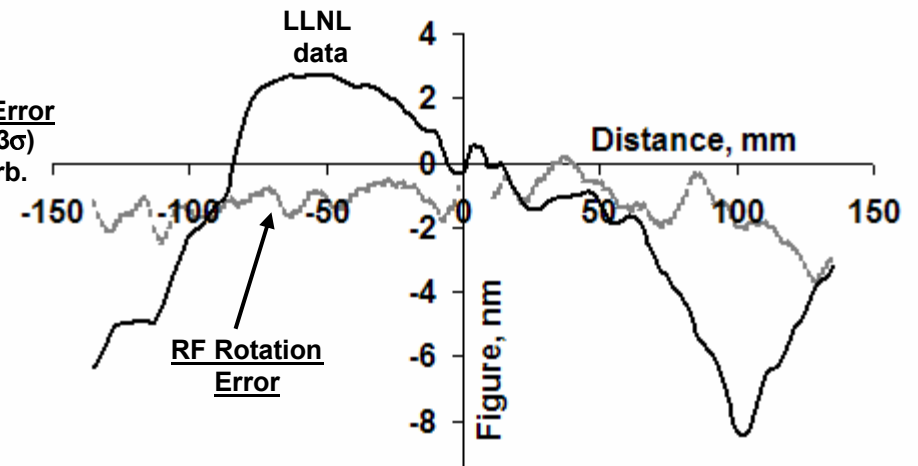
- Zygo does not calibrate the horizontal axis of “large” transmission flats (TF)
 - requires rotating a TF about the horizontal, which could break glass/epoxy bond
 - by design, 12” TF mounting cell design precludes this rotation
- Our horizontal calibration was done using two TF’s & the un-silvered side of an RF
 - a reflection flat (RF) mounting cell allows all three rotations (no glass/epoxy bond)

Vertical Calibration

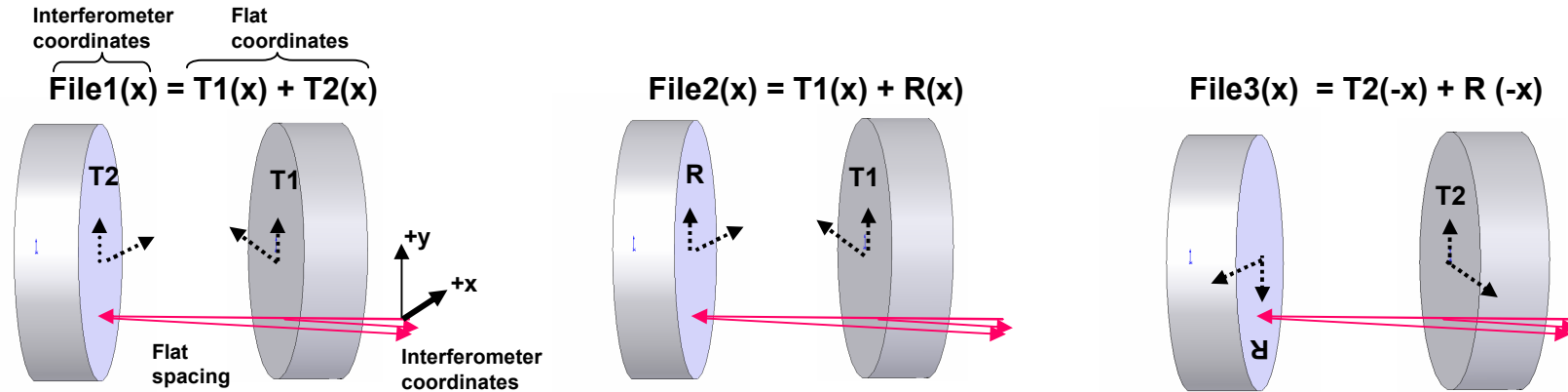


Random Error
 ± 2 nm (3σ)
 vib. + turb.

Horizontal Calibration



Algorithm for calibrating the horizontal axis of large flats



- Convert $\text{File3}(x) \rightarrow \text{File3}(-x)$ and solve for $T1(x)$, $T2(x)$, $R(x)$:

$$T1(x) = \frac{1}{2}[\text{File1}(x) + \text{File2}(x) - \text{File3}(-x)]$$

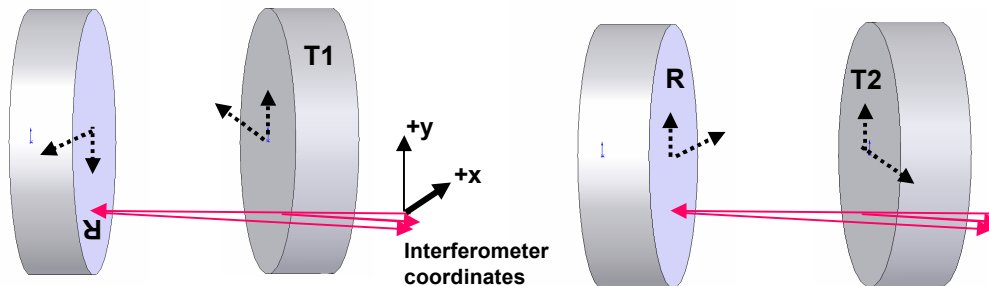
$$T2(x) = \frac{1}{2}[\text{File1}(x) - \text{File2}(x) + \text{File3}(-x)]$$

$$R(x) = \frac{1}{2}[-\text{File1}(x) + \text{File2}(x) + \text{File3}(-x)]$$

- The systematic error from RF weight reversal can be evaluated

$$\text{File4}(x) = T1(x) + [R(-x) + \text{Error}(-x)]$$

$$\text{File 5}(x) = T2(-x) + [R(x) + \text{Error}(x)]$$



Check
 $\text{Error}(x) = \text{File4}(-x) - T1(-x) - R(x)$
and double check
 $\text{Error}(x) = \text{File5}(x) - T2(-x) - R(x)$

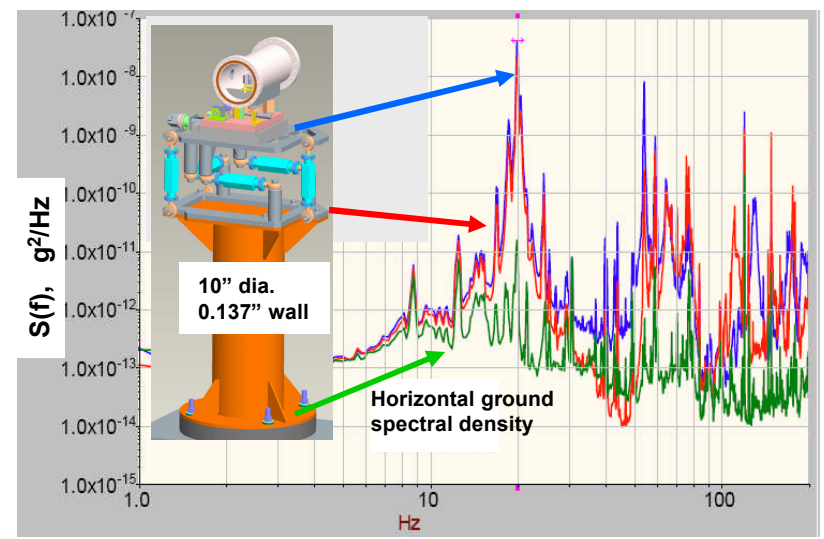
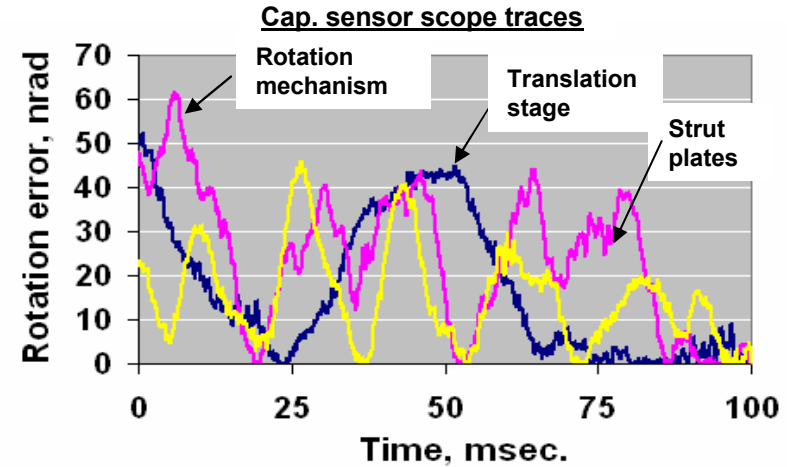


Effect of ground motion amplification on pointing stability has been addressed

- Pointing jitter was noticed on our first HOMS prototype
- Motion amplification was measured with accelerometers
 - rms amplification relates to mean square acceleration $S(f)$ by:
- The final design corrects this problem

$$\langle x_{top} \rangle_{rms} / \langle x_{ground} \rangle_{rms} = \sqrt{S_{top}(f) / S_{ground}(f)}$$

≈ 150 at 20 Hz



The final design fundamental frequency is over 80 Hz

- Modes calculated by a finite element model

PDR design

Mode	Hz	%Mass participation		
		X	Y	Z
1	27	66	0	0
2	32	0	0	74
3	80	3	3	0
4	97	1	53	0
5	100	12	2	0

FDR design

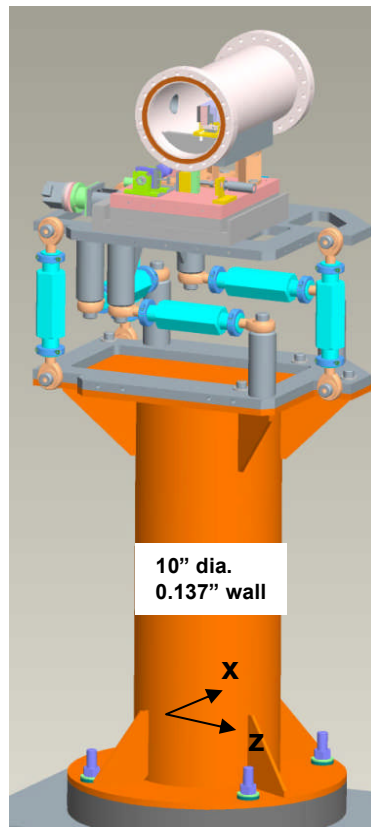
Mode	Hz	% Mass Participation		
		X	Y	Z
1	83	2	0	63
2	83	64	0	2
3	190	1	37	0
4	249	0	0	6
5	313	5	8	0

Amplitude “scaling” for mode f_n

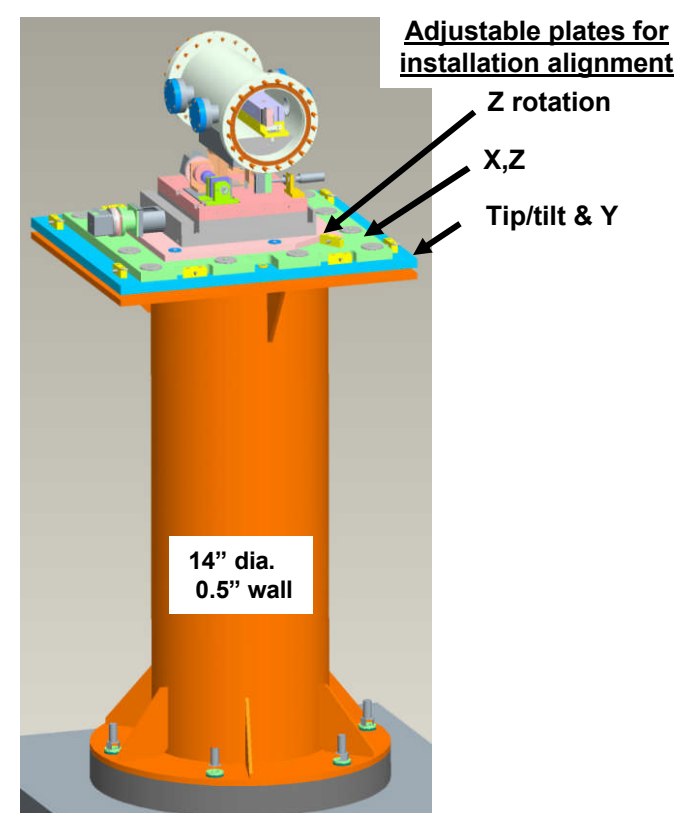
$$\langle x \rangle_{rms} = \frac{\left[\frac{\pi f_n S(f_n)}{\zeta} \right]^{1/2}}{2k}$$

← damping
← stiffness

PDR design



FDR design



Temperature induced pointing error has been measured and controlled

(1) Beam pointing drift < 10% of diameter required during experiments:

- $d\theta = 90$ nrad for HOMS
- $d\theta = 900$ nrad for SOMS

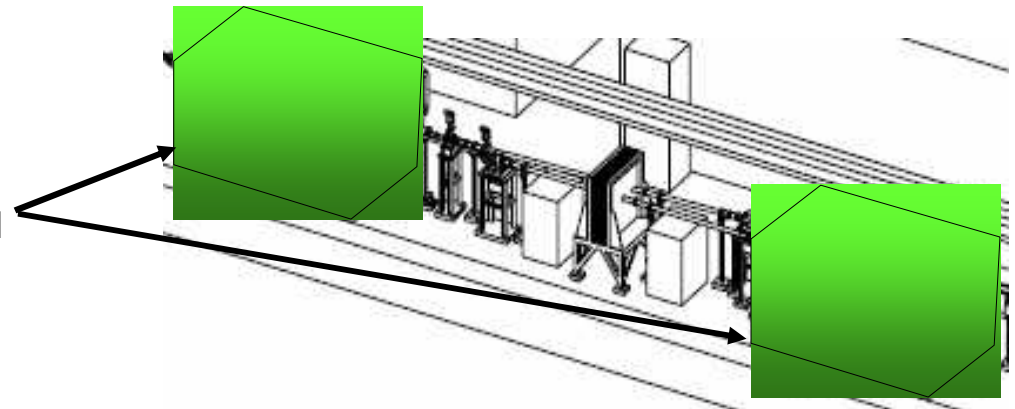
(2) Temperature dependence of mirror pointing has been measured: (next page)

- $d\theta/dT \cong 300$ nrad/0.1 °C

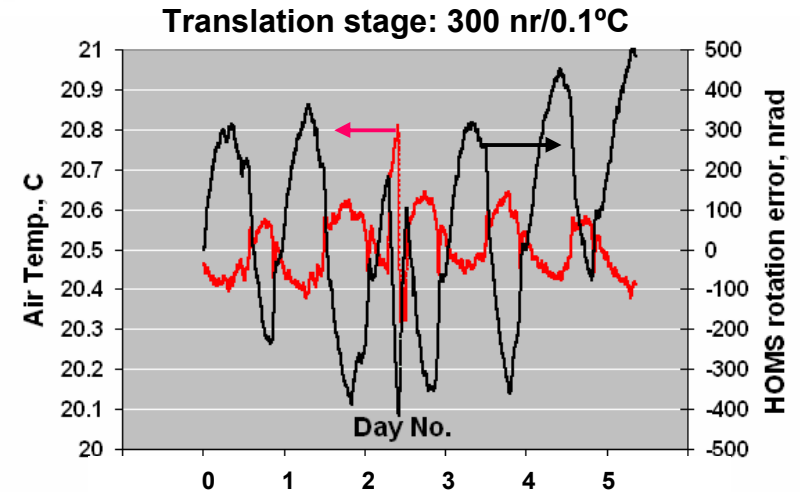
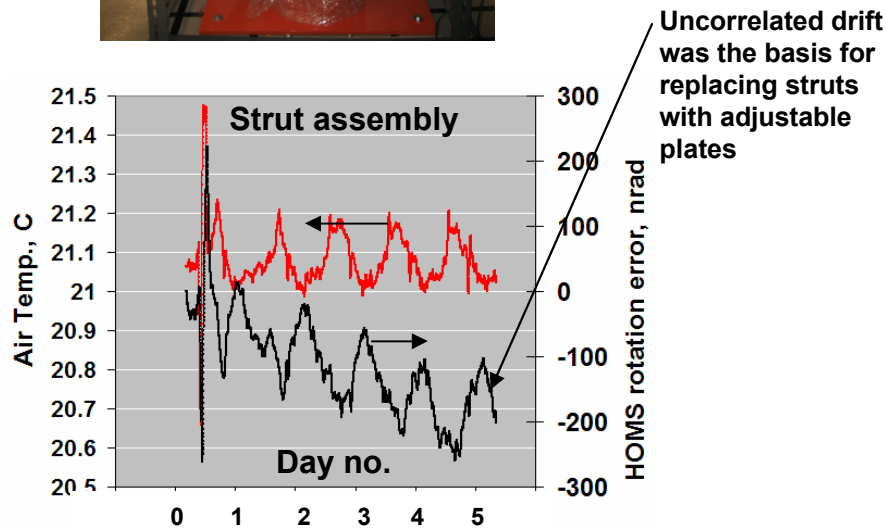
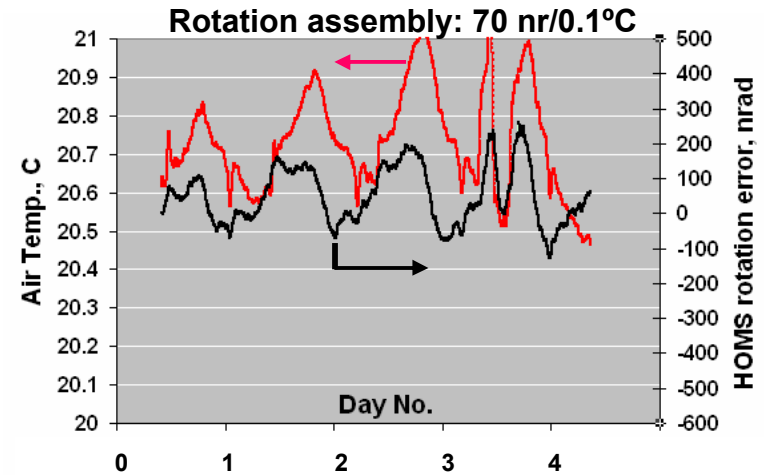
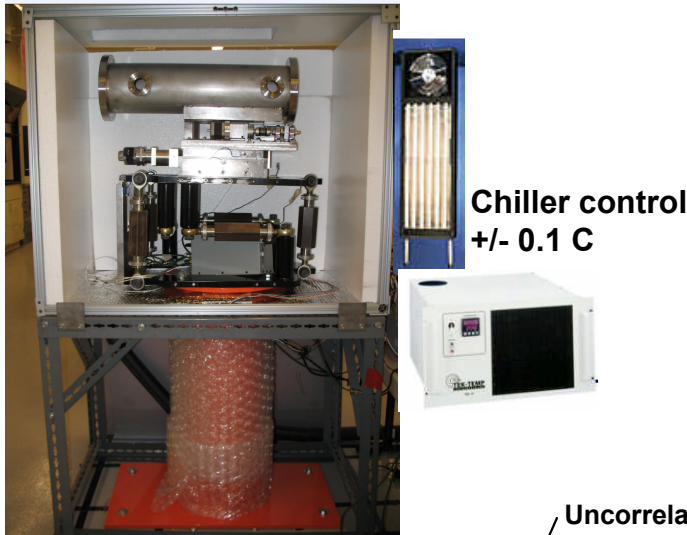
∴ Temperature stability < 0.03 C required for HOMS

(3) The challenge: to demonstrate closed loop stability < 0.03 C/week

- 0.01 C demonstrated in an insulated enclosure around our prototype



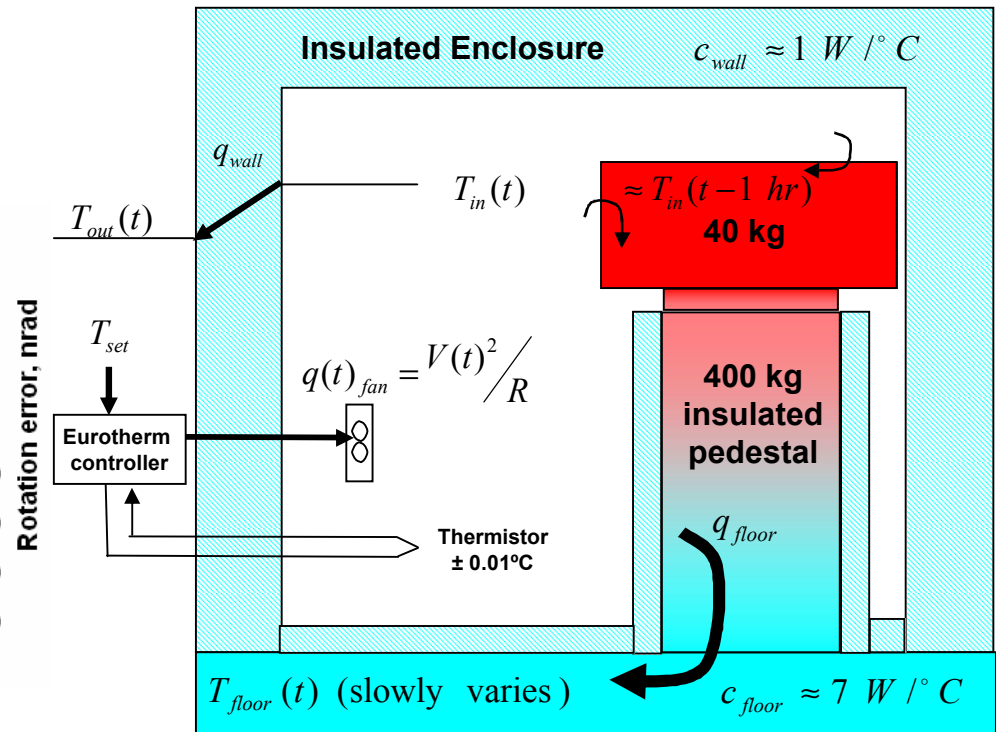
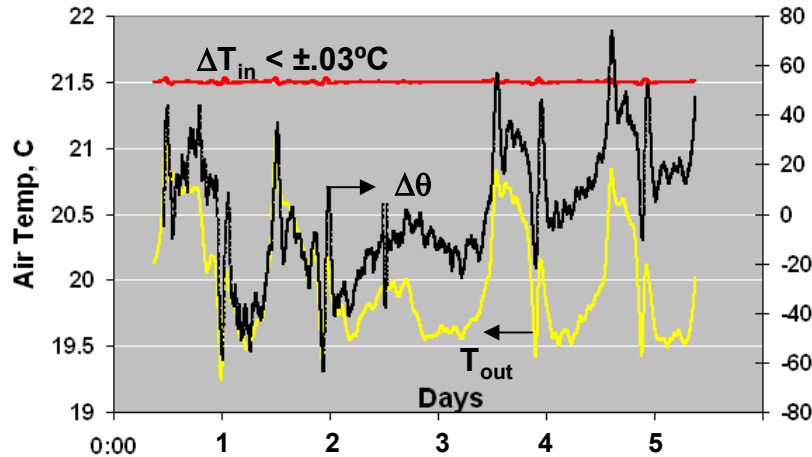
Pointing error $\Delta\theta/\Delta T$ was measured with our prototype



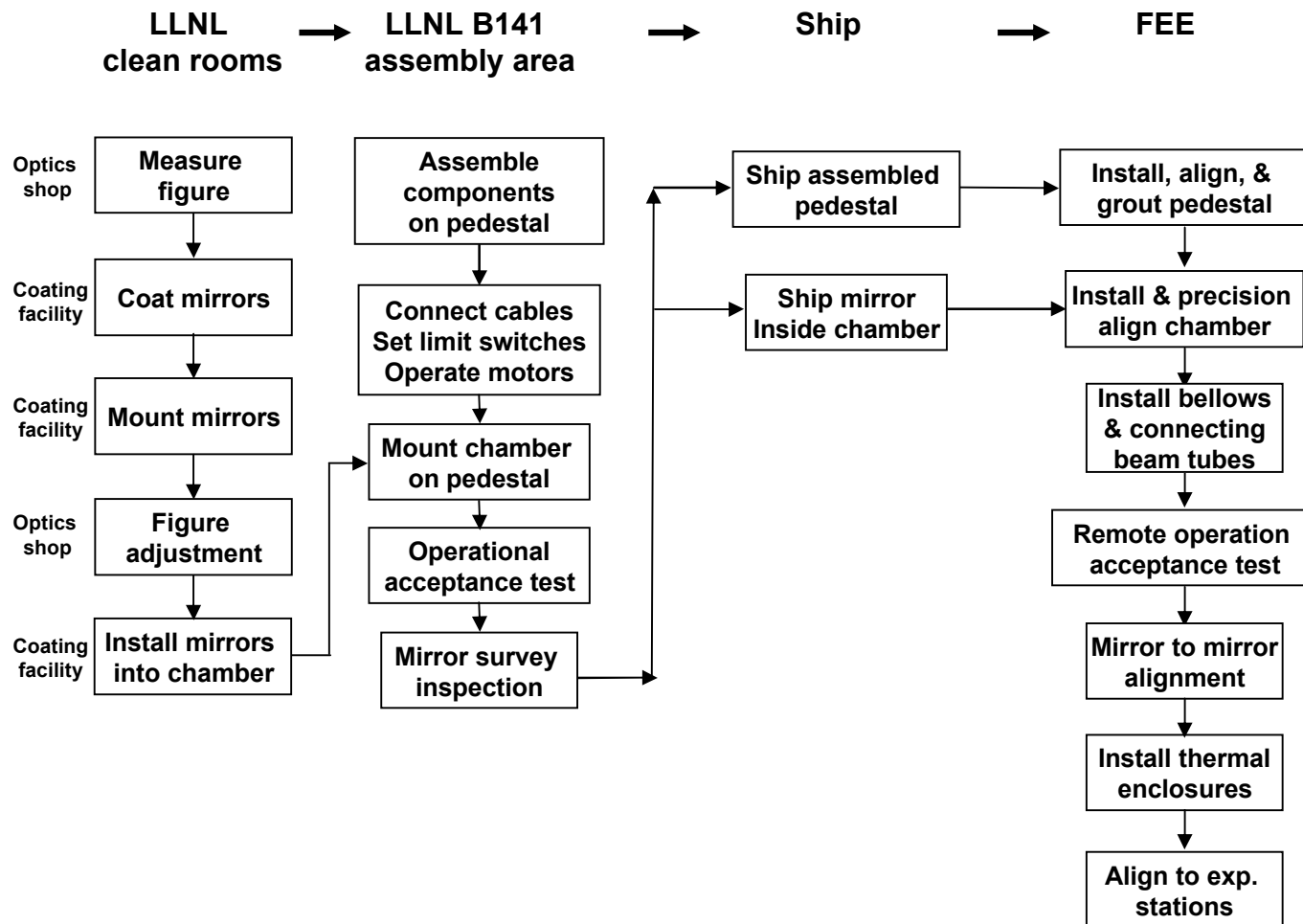
Pointing stability has been demonstrated in a thermal enclosure

- **Heat balance:** $q_{fan}(t) = q_{wall} + q_{floor} \approx c_{wall}[T_{in}(t) - T_{out}(t)] + c_{floor}[T_{in}(t-1 \text{ hr}) - T_{floor}(t)]$
 $dq_{fan}(t) \approx -c_{wall}dT_{out}(t)$

- **Control parameters:** $T_{set} = 21.5^\circ\text{C}$
 $q_{max} \approx 15 \text{ Watts}$



Summary of a mirror system “traveler”



Tasks remaining this FY:

- **Complete ESD, fabrication drawings, & final review – August**
 - calculate survey installation coordinates
- **Measure & coat SOMS mirrors – in August**
- **Bid and award fabrication drawings – Sept.**
 - (1) chamber
 - (2) pedestal & slide plate assembly
 - (3) mirror mount & chin guard assembly
 - (4) cam assembly

